THREE-DIMENSIONAL PROBLEM OF THE THEORY OF ELASTICITY STRESS IN A THICK-WALLED PRESSURE VESSEL

1. INTRODUCTION

Three-dimensional problem of the theory of elasticity includes an elastic body with defined kinematic or static boundary conditions and the mass forces acting inside. The analytical solution is known only for simple cases. In general, numerical methods are the only way to solve such tasks. Numerical solution of the problem by using FEM requires a three-dimensional spatial discretization with a solid three-dimensional finite elements.

2. PROBLEM DESCRIPTION

The goal of analysis is to determine stress distribution inside a pressure vessel made of steel which is a part of hydraulic installation. The vessel is loaded with internal pressure p. The vessel is attached by two flanges. The other two nozzles are free of displacements.

Data: p=50MPa, $E=2.10^5 MPa$, v=0.3Geometric data (in millimeters) are presented below:



3. TYPICAL COURSE OF NUMERICAL ANALYSIS

Taking into consideration the triple symmetry (xz, yz and zx planes), the model includes only $\frac{1}{8}$ part of the vessel. Convenient units are: *mm*, *N* and *MPa*.

3.1. Preprocessor









10. Select Element types: **Preprocessor**>Element Type>Add> (*SOLID186 or SOLID185*)









3.2. Solution

Define boundary conditions:







3.3. General postprocessor

Show the results as contour maps:

Show total displacements (USUM), Von Mises stress (SEQV) and stress components (SX, SY) in global cylindrical system related to cylindrical part of the model.



24. Select global cylindrical CS for results presentation:

ANSYS Main Menu	۲	1		
General Postproc	∧ Options for Output			×
Data & File Opts	Options for Output			
Results Summary	[RSYS] Results coord system		Global cylindric	-
Read Results	Local system reference no.			
Failure Criteria				
Plot Results	[AVPRIN] Principal stress calcs		From components	•
Deformed Shape Contour Plot	[AVRES] Avg rslts (pwr grph) for		All but Mat Prop	•
Nodal Solu	Use interior data		□ NO	
Element Solu	[/EFACET] Facets/element edge		1 facet/edge	-
Elem Table			T roccy cogo	
Line Elem Res	[SHELL] Shell results are from		- DEFAULT -	•
Vector Plot	[LAYER] Layer results are from			
Plot Path Item			Max failure crit	
Concrete Plot			Specified layer	
ThinFilm	Specified layer number		0	
List Results				_
Query Results	[FORCE] Force results are		Total force	•
Options for Outp				
Results Viewer				
Write PGR File				
Nodal Calcs	or	Cancel	Help	
Beinent Table		Cancer	help	

25. Plot radial stresses in global cylindrical CS (*RSYS=1*)









29. List radial and hoop stresses along **path AB**:









4. INTERPRETATION OF THE RESULTS. TASKS TO BE DONE

Compare results of the models built with the same mesh density (ESIZE parameter see p.12) using:

- a) 20-noded elements (Solid1865) using 'sweepping' HEX/WEDGE option (Model 1),
- b) 8-noded elements (Solid185) using 'sweepping' HEX/WEDGE option (Model 2),
- c) 8-noded elements (Solid185) using 'free meshing' TETRA option (Model 3).

Put the results in the **table** for each model:

No. of nodes, No. of elements, USUM_{max}, SEQV_{max}, SX_{RSYS=1}, SY_{RSYS=1} for points: A,B,C i D and maximum Membrane and Bending SEQV stress on path EF (step 35).

Discuss the results.

	Model 1 Solid186	Model 2 Solid185	Model 3 Solid185			
	Hex/Wed	Hex/Wed	Free			
No. of nodes				Plots needed (should be archived during program session for each model) :		
No. of elements				1) FE mesh		
				2) USUM(x,y)		
USUM _{max}				3) SEQV(x,y)		
SEQV _{max}				4) SX(x,y) _{RSYS=1}		
CV/4				5) SY(x,y) _{RSYS=1}		
SX ^A RSYS=1				6) Graph: $SX(x,y)_{RSYS=1}$ i $SY(x,y)_{RSYS=1}$ on path AB		
SYA RSYS=1				7) Graph: SX(x,y) _{RSYS=1} i SY(x,y) _{RSYS=1} on path CD		
				8) Graph of linearized SEQV on path EF		
SX ^B _{RSYS=1}						
SY ^B _{RSYS=1}				Raport finalny:		
SX ^C _{RSYS=1}				Final report:		
SY ^C _{RSYS=1}				 Assumptions for the modeling model description 		
SX ^D RSYS=1				4) Results 5) Results in the Table		
SYD RSYS=1				6) Discursion 7) Conclusion		
Max Membrane + Bending stress						
from Lame theorem (for inside pressure):						
$\sigma_r = \frac{p_a \cdot a^2}{b^2 - a^2} \cdot (1 - \frac{b^2}{r^2}) \qquad \qquad \sigma_t = \frac{p_a \cdot a^2}{b^2 - a^2} \cdot (1 + \frac{b^2}{r^2})$			$\sigma_t \sigma_r$			
$\sigma_r(a) \equiv$			$\sigma_t(b)$			
$\sigma_{i}(a) =$			VITE KIA			
$o_t(u) =$						
$\sigma_r(b) =$				$\sigma_r(a)$		
$\sigma_t(h) \equiv$						
(())—						